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## Introduction, Classification, and System

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### 1-1C

**Solution** We are to define a fluid and how it differs between a solid and a gas.

**Analysis** A substance in the liquid or gas phase is referred to as a fluid. A fluid differs from a solid in that a solid can resist an applied shear stress by deforming, whereas a fluid deforms continuously under the influence of shear stress, no matter how small. A liquid takes the shape of the container it is in, and a liquid forms a free surface in a larger container in a gravitational field. A gas, on the other hand, expands until it encounters the walls of the container and fills the entire available space.

**Discussion** The subject of fluid mechanics deals with all fluids, both gases and liquids.

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### 1-2C

**Solution** We are to define internal, external, and open-channel flows.

**Analysis** External flow is the flow of an unbounded fluid over a surface such as a plate, a wire, or a pipe. The flow in a pipe or duct is internal flow if the fluid is completely bounded by solid surfaces. The flow of liquids in a pipe is called open-channel flow if the pipe is partially filled with the liquid and there is a free surface, such as the flow of water in rivers and irrigation ditches.

**Discussion** As we shall see in later chapters, different approximations are used in the analysis of fluid flows based on their classification.

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### 1-3C

**Solution** We are to define incompressible and compressible flow, and discuss fluid compressibility.

**Analysis** A fluid flow during which the density of the fluid remains nearly constant is called incompressible flow. A flow in which density varies significantly is called compressible flow. A fluid whose density is practically independent of pressure (such as a liquid) is commonly referred to as an “incompressible fluid,” although it is more proper to refer to incompressible flow. The flow of compressible fluid (such as air) does not necessarily need to be treated as compressible since the density of a compressible fluid may still remain nearly constant during flow – especially flow at low speeds.

**Discussion** It turns out that the Mach number is the critical parameter to determine whether the flow of a gas can be approximated as an incompressible flow. If  $Ma$  is less than about 0.3, the incompressible approximation yields results that are in error by less than a couple percent.

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1-4C

**Solution** We are to determine whether the flow of air over the wings of an aircraft and the flow of gases through a jet engine is internal or external.

**Analysis** The flow of air over the wings of an aircraft is **external** since this is an unbounded fluid flow over a surface. The flow of gases through a jet engine is **internal** flow since the fluid is completely bounded by the solid surfaces of the engine.

**Discussion** If we consider the entire airplane, the flow is both internal (through the jet engines) and external (over the body and wings).

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1-5C

**Solution** We are to define forced flow and discuss the difference between forced and natural flow. We are also to discuss whether wind-driven flows are forced or natural.

**Analysis** In *forced flow*, the fluid is forced to flow over a surface or in a tube **by external means** such as a pump or a fan. In *natural flow*, any fluid motion is caused by natural means such as the buoyancy effect that manifests itself as the rise of the warmer fluid and the fall of the cooler fluid. **The flow caused by winds is natural flow for the earth, but it is forced flow for bodies subjected to the winds** since for the body it makes no difference whether the air motion is caused by a fan or by the winds.

**Discussion** As seen here, the classification of forced vs. natural flow may depend on your frame of reference.

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1-6C

**Solution** We are to define the Mach number of a flow and the meaning for a Mach number of 2.

**Analysis** The Mach number of a flow is defined as **the ratio of the speed of flow to the speed of sound** in the flowing fluid. **A Mach number of 2 indicate a flow speed that is twice the speed of sound in that fluid.**

**Discussion** Mach number is an example of a dimensionless (or nondimensional) parameter.

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1-7C

**Solution** We are to discuss if the Mach number of a constant-speed airplane is constant.

**Analysis** No. The speed of sound, and thus the Mach number, changes with temperature which may change considerably from point to point in the atmosphere.

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1-8C

**Solution** We are to determine if the flow of air with a Mach number of 0.12 should be approximated as incompressible.

**Analysis** Gas flows can often be approximated as incompressible if the density changes are under about 5 percent, which is usually the case when  $Ma < 0.3$ . Therefore, air flow with a Mach number of 0.12 **may be approximated as being incompressible**.

**Discussion** Air is of course a compressible fluid, but at low Mach numbers, compressibility effects are insignificant.

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1-9C

**Solution** We are to define the no-slip condition and its cause.

**Analysis** A **fluid in direct contact with a solid surface sticks to the surface and there is no slip**. This is known as the *no-slip condition*, and it is due to the *viscosity* of the fluid.

**Discussion** There is no such thing as an inviscid fluid, since all fluids have viscosity.

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1-10C

**Solution** We are to define a boundary layer, and discuss its cause.

**Analysis** The **region of flow** (usually near a wall) **in which the velocity gradients are significant and frictional effects are important** is called the *boundary layer*. When a fluid stream encounters a solid surface that is at rest, the fluid velocity assumes a value of zero at that surface. The velocity then varies from zero at the surface to some larger value sufficiently far from the surface. **The development of a boundary layer is caused by the *no-slip condition*.**

**Discussion** As we shall see later, flow within a boundary layer is *rotational* (individual fluid particles rotate), while that outside the boundary layer is typically *irrotational* (individual fluid particles move, but do not rotate).

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1-11C

**Solution** We are to define a steady-flow process.

**Analysis** A process is said to be *steady* if it involves **no changes with time** anywhere within the system or at the system boundaries.

**Discussion** The opposite of steady flow is *unsteady flow*, which involves changes with time.

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1-12C

**Solution** We are to define stress, normal stress, shear stress, and pressure.

**Analysis** *Stress* is defined as **force per unit area**, and is determined by dividing the force by the area upon which it acts. The **normal component of a force acting on a surface per unit area** is called the *normal stress*, and the **tangential component of a force acting on a surface per unit area** is called *shear stress*. In a fluid at rest, the normal stress is called *pressure*.

**Discussion** Fluids in motion may have both shear stresses and additional normal stresses besides pressure, but when a fluid is at rest, the only normal stress is the pressure, and there are no shear stresses.

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1-13C

**Solution** We are to define system, surroundings, and boundary.

**Analysis** A *system* is defined as a **quantity of matter or a region in space chosen for study**. The mass or **region outside the system** is called the *surroundings*. The real or imaginary **surface that separates the system from its surroundings** is called the *boundary*.

**Discussion** Some authors like to define *closed systems* and *open systems*, while others use the notation “system” to mean a closed system and “control volume” to mean an open system. This has been a source of confusion for students for many years.

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1-14C

**Solution** We are to discuss how to select system when analyzing the acceleration of gases as they flow through a nozzle.

**Analysis** When analyzing the acceleration of gases as they flow through a nozzle, a wise choice for the system is **the volume within the nozzle**, bounded by the entire inner surface of the nozzle and the inlet and outlet cross-sections. This is a **control volume (or open system)** since mass crosses the boundary.

**Discussion** It would be much more difficult to follow a chunk of air as a closed system as it flows through the nozzle.

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1-15C

**Solution** We are to discuss when a system is considered closed or open.

**Analysis** Systems may be considered to be *closed* or *open*, depending on whether a fixed mass or a volume in space is chosen for study. A *closed system* (also known as a *control mass* or simply a *system*) consists of a **fixed amount of mass, and no mass can cross its boundary**. An *open system*, or a *control volume*, is a **selected region in space**. Mass may cross the boundary of a control volume or open system.

**Discussion** In thermodynamics, it is more common to use the terms *open system* and *closed system*, but in fluid mechanics, it is more common to use the terms *system* and *control volume* to mean the same things, respectively.

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1-16C

**Solution** We are to discuss how to select system for the operation of a reciprocating air compressor.

**Analysis** We would most likely take the system as **the air contained in the piston-cylinder device**. This system is a **closed or fixed mass system when it is compressing and no mass enters or leaves it**. However, it is an **open system during intake or exhaust**.

**Discussion** In this example, the system boundary is the same for either case – closed or open system.

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**Mass, Force, and Units**

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**1-17C**

**Solution** We are to discuss the difference between pound-mass and pound-force.

**Analysis** *Pound-mass* lbm is the **mass unit in English system** whereas *pound-force* lbf is the **force unit in the English system**. One pound-force is the force required to accelerate a mass of 32.174 lbm by  $1 \text{ ft/s}^2$ . In other words, the weight of a 1-lbm mass at sea level on earth is 1 lbf.

**Discussion** It is *not* proper to say that one lbm is equal to one lbf since the two units have different dimensions.

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**1-18C**

**Solution** We are to discuss the difference between pound-mass (lbm) and pound-force (lbf).

**Analysis** The “pound” mentioned here must be “**lbf**” since thrust is a force, and the lbf is the force unit in the English system.

**Discussion** You should get into the habit of *never* writing the unit “lb”, but always use either “lbm” or “lbf” as appropriate since the two units have different dimensions.

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**1-19C**

**Solution** We are to explain why the light-year has the dimension of length.

**Analysis** In this unit, the word *light* refers to the speed of light. The light-year unit is then the product of a velocity and time. Hence, this product forms a distance dimension and unit.

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**1-20C**

**Solution** We are to calculate the net force on a car cruising at constant velocity.

**Analysis** There is no acceleration (car moving at constant velocity), thus **the net force is zero in both cases**.

**Discussion** By Newton’s second law, the force on an object is directly proportional to its acceleration. If there is zero acceleration, there must be zero net force.

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## 1-21

**Solution** A man is considering buying a 12-oz steak for \$3.15, or a 320-g steak for \$3.30. The steak that is a better buy is to be determined.

**Assumptions** The steaks are of identical quality.

**Analysis** To make a comparison possible, we need to express the cost of each steak on a common basis. We choose 1 kg as the basis for comparison. Using proper conversion factors, the unit cost of each steak is determined to be

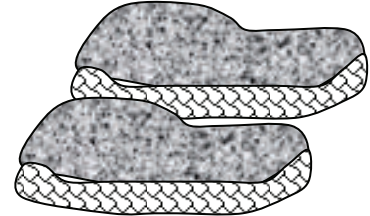
$$12 \text{ ounce steak: Unit Cost} = \left( \frac{\$3.15}{12 \text{ oz}} \right) \left( \frac{16 \text{ oz}}{1 \text{ lbm}} \right) \left( \frac{1 \text{ lbm}}{0.45359 \text{ kg}} \right) = \mathbf{\$9.26/\text{kg}}$$

320 gram steak:

$$\text{Unit Cost} = \left( \frac{\$3.30}{320 \text{ g}} \right) \left( \frac{1000 \text{ g}}{1 \text{ kg}} \right) = \mathbf{\$10.3/\text{kg}}$$

Therefore, **the steak at the traditional market is a better buy.**

**Discussion** Notice the unity conversion factors in the above equations.



## 1-22

**Solution** The mass of an object is given. Its weight is to be determined.

**Analysis** Applying Newton's second law, the weight is determined to be

$$W = mg = (150 \text{ kg})(9.6 \text{ m/s}^2) = \mathbf{1440 \text{ N}}$$

## 1-23

**Solution** The mass of a substance is given. Its weight is to be determined in various units.

**Analysis** Applying Newton's second law, the weight is determined in various units to be

$$W = mg = (1 \text{ kg})(9.81 \text{ m/s}^2) \left( \frac{1 \text{ N}}{1 \text{ kg} \cdot \text{m/s}^2} \right) = \mathbf{9.81 \text{ N}}$$

$$W = mg = (1 \text{ kg})(9.81 \text{ m/s}^2) \left( \frac{1 \text{ kN}}{1000 \text{ kg} \cdot \text{m/s}^2} \right) = \mathbf{0.00981 \text{ kN}}$$

$$W = mg = (1 \text{ kg})(9.81 \text{ m/s}^2) = \mathbf{1 \text{ kg} \cdot \text{m/s}^2}$$

$$W = mg = (1 \text{ kg})(9.81 \text{ m/s}^2) \left( \frac{1 \text{ N}}{1 \text{ kg} \cdot \text{m/s}^2} \right) \left( \frac{1 \text{ kgf}}{9.81 \text{ N}} \right) = \mathbf{1 \text{ kgf}}$$

$$W = mg = (1 \text{ kg}) \left( \frac{2.205 \text{ lbm}}{1 \text{ kg}} \right) (32.2 \text{ ft/s}^2) = \mathbf{71 \text{ lbm} \cdot \text{ft/s}^2}$$

$$W = mg = (1 \text{ kg}) \left( \frac{2.205 \text{ lbm}}{1 \text{ kg}} \right) (32.2 \text{ ft/s}^2) \left( \frac{1 \text{ lbf}}{32.2 \text{ lbm} \cdot \text{ft/s}^2} \right) = \mathbf{2.21 \text{ lbf}}$$

## 1-24

**Solution** The interior dimensions of a room are given. The mass and weight of the air in the room are to be determined.

**Assumptions** The density of air is constant throughout the room.

**Properties** The density of air is given to be  $\rho = 1.16 \text{ kg/m}^3$ .

**Analysis** The mass of the air in the room is

$$m = \rho V = (1.16 \text{ kg/m}^3)(3 \times 5 \times 7 \text{ m}^3) = 121.8 \text{ kg} \cong \mathbf{122 \text{ kg}}$$

Thus,

$$W = mg = (121.8 \text{ kg})(9.81 \text{ m/s}^2) \left( \frac{1 \text{ N}}{1 \text{ kg} \cdot \text{m/s}^2} \right) = \mathbf{1195 \text{ N}}$$

Room air

$$3 \times 5 \times 7 \text{ m}^3$$

**Discussion** Note that we round our final answers to three or four significant digits, but use extra digit(s) in intermediate calculations. Considering that the mass of an average man is about 70 to 90 kg, the mass of air in the room is probably larger than you might have expected.

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## 1-25

**Solution** A resistance heater is used to heat water to desired temperature. The amount of electric energy used in kWh and kJ are to be determined.

**Analysis** The resistance heater consumes electric energy at a rate of 3 kW or 3 kJ/s. Then the total amount of electric energy used in 2 hours becomes

$$\begin{aligned} \text{Total energy} &= (\text{Energy per unit time})(\text{Time interval}) \\ &= (3 \text{ kW})(2 \text{ h}) \\ &= \mathbf{6 \text{ kWh}} \end{aligned}$$

Noting that  $1 \text{ kWh} = (1 \text{ kJ/s})(3600 \text{ s}) = 3600 \text{ kJ}$ ,

$$\begin{aligned} \text{Total energy} &= (6 \text{ kWh})(3600 \text{ kJ/kWh}) \\ &= \mathbf{21,600 \text{ kJ}} \end{aligned}$$

**Discussion** Note kW is a unit for power whereas kWh is a unit for energy.

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## 1-26E

**Solution** An astronaut takes his scales with him to the moon. It is to be determined how much he weighs on the spring and beam scales on the moon.

**Analysis**

(a) A spring scale measures weight, which is the local gravitational force applied on a body:

$$W = mg = (195 \text{ lbm})(5.48 \text{ ft/s}^2) \left( \frac{1 \text{ lbf}}{32.2 \text{ lbm} \cdot \text{ft/s}^2} \right) = \mathbf{33.2 \text{ lbf}}$$

(b) A beam scale compares masses and thus is not affected by the variations in gravitational acceleration. The beam scale reads what it reads on earth,

$$W = \mathbf{195 \text{ lbf}}$$

**Discussion** The beam scale may be marked in units of weight (lbf), but it really compares mass, not weight. Which scale would you consider to be more accurate?

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## 1-27

**Solution** The acceleration of an aircraft is given in  $g$ 's. The net upward force acting on a man in the aircraft is to be determined.

**Analysis** From Newton's second law, the applied force is

$$F = ma = m(6g) = (90 \text{ kg})(6 \times 9.81 \text{ m/s}^2) \left( \frac{1 \text{ N}}{1 \text{ kg} \cdot \text{m/s}^2} \right) = 5297 \text{ N} \cong \mathbf{5300 \text{ N}}$$

where we have rounded off the final answer to three significant digits.

**Discussion** The man feels like he is six times heavier than normal. You get a similar feeling when riding an elevator to the top of a tall building, although to a much lesser extent.

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## 1-28

**Solution** A rock is thrown upward with a specified force. The acceleration of the rock is to be determined.

**Analysis** The weight of the rock is

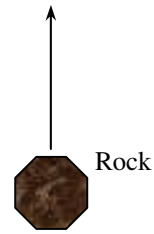
$$W = mg = (10 \text{ kg})(9.79 \text{ m/s}^2) \left( \frac{1 \text{ N}}{1 \text{ kg} \cdot \text{m/s}^2} \right) = \mathbf{97.9 \text{ N}}$$

Then the net force that acts on the rock is

$$F_{\text{net}} = F_{\text{up}} - F_{\text{down}} = 280 - 97.9 = 182.1 \text{ N}$$

From Newton's second law, the acceleration of the rock becomes

$$a = \frac{F}{m} = \frac{182.1 \text{ N}}{10 \text{ kg}} \left( \frac{1 \text{ kg} \cdot \text{m/s}^2}{1 \text{ N}} \right) = \mathbf{18.2 \text{ m/s}^2}$$



**Discussion** This acceleration is more than twice the acceleration at which it would fall (due to gravity) if dropped.

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1-29 

**Solution** The previous problem is reconsidered. The entire software solution is to be printed out, including the numerical results with proper units.

**Analysis** The problem is solved using EES (Engineering Equation Solver), and the solution is given below.

```
m=10 [kg]
F_up=280 [N]
g=9.79 [m/s^2]
W=m*g
F_net = F_up - F_down
F_down=W
F_net=a*m
```

```
SOLUTION
a=18.21 [m/s^2]
F_down=97.9 [N]
F_net=182.1 [N]
F_up=280 [N]
g=9.79 [m/s^2]
m=10 [kg]
W=97.9 [N]
```

The final results are  $W = 97.9 \text{ N}$  and  $a = 18.2 \text{ m/s}^2$ , to three significant digits, which agree with the results of the previous problem.

**Discussion** Items in quotation marks in the EES Equation window are comments. Units are in square brackets.

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## 1-30

**Solution** Gravitational acceleration  $g$  and thus the weight of bodies decreases with increasing elevation. The percent reduction in the weight of an airplane cruising at 13,000 m is to be determined.

**Properties** The gravitational acceleration  $g$  is  $9.807 \text{ m/s}^2$  at sea level and  $9.767 \text{ m/s}^2$  at an altitude of 13,000 m.

**Analysis** Weight is proportional to the gravitational acceleration  $g$ , and thus the percent reduction in weight is equivalent to the percent reduction in the gravitational acceleration, which is determined from

$$\% \text{ Reduction in weight} = \% \text{ Reduction in } g = \frac{\Delta g}{g} \times 100 = \frac{9.807 - 9.767}{9.807} \times 100 = \mathbf{0.41\%}$$

Therefore, **the airplane and the people in it will weigh 0.41% less at 13,000 m altitude.**

**Discussion** Note that the weight loss at cruising altitudes is negligible. Sorry, but flying in an airplane is not a good way to lose weight. The best way to lose weight is to carefully control your diet, and to exercise.

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## 1-31

**Solution** The variation of gravitational acceleration above sea level is given as a function of altitude. The height at which the weight of a body decreases by 1% is to be determined.

**Analysis** The weight of a body at the elevation  $z$  can be expressed as

$$W = mg = m(a - bz)$$

where  $a = g_s = 9.807 \text{ m/s}^2$  is the value of gravitational acceleration at sea level and  $b = 3.32 \times 10^{-6} \text{ s}^{-2}$ .

In our case,

$$W = m(a - bz) = 0.99W_s = 0.99mg_s$$

We cancel out mass from both sides of the equation and solve for  $z$ , yielding

$$z = \frac{a - 0.99g_s}{b}$$

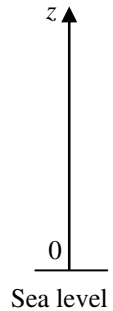
Substituting,

$$z = \frac{9.807 \text{ m/s}^2 - 0.99(9.807 \text{ m/s}^2)}{3.32 \times 10^{-6} \text{ 1/s}^2} = 29,539 \text{ m} \cong \mathbf{29,500 \text{ m}}$$

where we have rounded off the final answer to three significant digits.

**Discussion** This is more than three times higher than the altitude at which a typical commercial jet flies, which is about 30,000 ft (9140 m). So, flying in a jet is not a good way to lose weight – diet and exercise are always the best bet.

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## 1-32

**Solution** A relation for the gravitational constant as a function elevation is given. The weight of an astronaut in the International Space Station is to be determined.

**Analysis** At the altitude of the Space Station,

$$g = 9.807 - (3.32 \times 10^{-6} \text{ s}^{-2})(354,000 \text{ m}) = 8.6317 \text{ m/s}^2$$

The astronaut's weight would therefore be

$$W = mg = (80.0 \text{ kg})(8.6317 \text{ m/s}^2) \left( \frac{\text{N}}{\text{kg} \cdot \text{m/s}^2} \right) = 690.538 \text{ N}$$

Or, rounding off,  $W = \mathbf{690.5 \text{ N}}$ . This is the astronaut's weight if he were at that elevation, but not in orbit.

However, since the satellite and its occupants are in orbit, **the astronaut feels weightless**. In other words, if the person were to step on a bathroom scale, the reading would be zero.

The astronaut feels a gravitational acceleration of  $\mathbf{8.93 \text{ m/s}^2}$ . But since the satellite and its occupants are in orbit, they are in free fall – constantly accelerating (falling) at this rate, but also moving horizontally at a speed such that the elevation of the satellite remains constant, and its circular orbit is maintained.

The astronaut feels weightless only because he or she is in a steady circular orbit around the earth. The gravitational acceleration on a satellite is in fact the centripetal acceleration (towards the earth, i.e., “down”) that maintains the satellite's circular orbit. The astronaut “feels” weightless while in orbit in the same way that a person jumping off a diving board “feels” weightless during free fall.

**Discussion** The astronaut's actual weight is only about 12% smaller than on the earth's surface! It is a common misconception that space has zero gravity. In fact, gravity does indeed decrease away from earth's surface, but it is still fairly strong even at altitudes where geosynchronous satellites orbit. In fact, if you think about it, satellites could not maintain an orbit at all without gravity acting on them. The simple linear equation we used here for gravitational decay with elevation breaks down at high elevation, and a more appropriate equation should be used.

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## 1-33

**Solution** The mass of air that a person breathes in per day is to be determined.

**Analysis** The total volume of air breathed in per day is

$$V = \left( 7.0 \frac{\text{L}}{\text{min}} \right) \left( 60 \frac{\text{min}}{\text{hr}} \right) \left( 24 \frac{\text{hr}}{\text{day}} \right) \left( \frac{1 \text{ m}^3}{1000 \text{ L}} \right) = 10.08 \text{ m}^3$$

We use the ideal gas law to calculate the total mass of air,

$$PV = mRT \quad \rightarrow \quad m = \frac{PV}{RT} = \frac{(101.325 \text{ kPa})(10.08 \text{ m}^3)}{\left( 0.287 \frac{\text{kJ}}{\text{kg} \cdot \text{K}} \right) (298.15 \text{ K})} \left( 1 \frac{\text{kN/m}^2}{\text{kPa}} \right) \left( 1 \frac{\text{kJ}}{\text{kN} \cdot \text{m}} \right) = 11.936 \text{ kg}$$

So, to two significant digits, **an average person breathes in about 12 kg of air per day, or about 17% of the average person's mass**.

**Discussion** This is a lot more air than you probably thought! We breathe in more air (in terms of mass) than the mass of food that we eat!

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## 1-34

**Solution** During an analysis, a relation with inconsistent units is obtained. A correction is to be found, and the probable cause of the error is to be determined.

**Analysis** The two terms on the right-hand side of the equation

$$E = 16 \text{ kJ} + 7 \text{ kJ/kg}$$

do not have the same units, and therefore they cannot be added to obtain the total energy. **Multiplying the last term by mass will eliminate the kilograms in the denominator, and the whole equation will become dimensionally homogeneous;** that is, every term in the equation will have the same unit.

**Discussion** Obviously this error was caused by forgetting to multiply the last term by mass at an earlier stage.

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## 1-35

**Solution** We are to calculate the useful power delivered by an airplane propeller.

**Assumptions** 1 The airplane flies at constant altitude and constant speed. 2 Wind is not a factor in the calculations.

**Analysis** At steady horizontal flight, the airplane's drag is balanced by the propeller's thrust. Energy is force times distance, and power is energy per unit time. Thus, by dimensional reasoning, the power supplied by the propeller must equal thrust times velocity,

$$\dot{W} = F_{\text{thrust}} V = (1500 \text{ N})(70.0 \text{ m/s}) \left( \frac{1 \text{ kW}}{1000 \text{ N} \cdot \text{m/s}} \right) = \mathbf{105 \text{ kW}} \left( \frac{1.341 \text{ hp}}{1 \text{ kW}} \right) = \mathbf{141 \text{ hp}}$$

where we give our final answers to 3 significant digits.

**Discussion** We used two unity conversion ratios in the above calculation. The actual shaft power supplied by the airplane's engine will of course be larger than that calculated above due to inefficiencies in the propeller.

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## 1-36

**Solution** We are to calculate lift produced by an airplane's wings.

**Assumptions** 1 The airplane flies at constant altitude and constant speed. 2 Wind is not a factor in the calculations.

**Analysis** At steady horizontal flight, the airplane's weight is balanced by the lift produced by the wings. Thus, the net lift force must equal the weight, or  $F_L = 1700 \text{ lbf}$ . We use unity conversion ratios to convert to newtons:

$$F_L = (1700 \text{ lbf}) \left( \frac{1 \text{ N}}{0.22481 \text{ lbf}} \right) = \mathbf{7560 \text{ N}}$$

where we give our final answers to 3 significant digits.

**Discussion** The answer is valid at any speed, since lift must balance weight in order to sustain straight, horizontal flight. As the fuel is consumed, the overall weight of the aircraft will decrease, and hence the lift requirement will also decrease. If the pilot does not adjust, the airplane will climb slowly in altitude.

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## 1-37E

**Solution** We are to estimate the work required to lift a fireman, and estimate how long it takes.

**Assumptions** 1 The vertical speed of the fireman is constant.

**Analysis**

(a) Work  $W$  is a form of energy, and is equal to force times distance. Here, the force is the weight of the fireman (and equipment), and the vertical distance is  $\Delta z$ , where  $z$  is the elevation.

$$W = F\Delta z = (280 \text{ lbf})(40.0 \text{ ft}) \left( \frac{1 \text{ Btu}}{778.169 \text{ ft} \cdot \text{lbf}} \right) = 14.393 \text{ Btu} \cong \mathbf{14.4 \text{ Btu}}$$

where we give our final answer to 3 significant digits, but retain 5 digits to avoid round-off error in part (b).

(b) Power is work (energy) per unit time. Assuming a constant speed,

$$\Delta t = \frac{W}{\dot{W}} = \frac{14.393 \text{ Btu}}{3.50 \text{ hp}} \left( \frac{1 \text{ hp}}{0.7068 \text{ Btu/s}} \right) = 5.8182 \text{ s} \cong \mathbf{5.82 \text{ s}}$$

Again we give our final answer to 3 significant digits.

**Discussion** The actual required power will be greater than calculated here, due to frictional losses and other inefficiencies in the boom's lifting system. One unity conversion ratio is used in each of the above calculations.

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## 1-38

**Solution** A plastic tank is filled with water. The weight of the combined system is to be determined.

**Assumptions** The density of water is constant throughout.

**Properties** The density of water is given to be  $\rho = 1000 \text{ kg/m}^3$ .

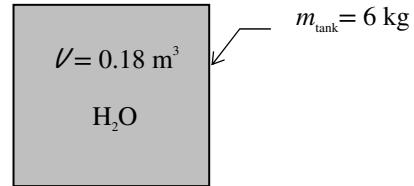
**Analysis** The mass of the water in the tank and the total mass are

$$m_w = \rho V = (1000 \text{ kg/m}^3)(0.18 \text{ m}^3) = 180 \text{ kg}$$

$$m_{\text{total}} = m_w + m_{\text{tank}} = 180 + 6 = 186 \text{ kg}$$

Thus,

$$W = mg = (186 \text{ kg})(9.81 \text{ m/s}^2) \left( \frac{1 \text{ N}}{1 \text{ kg} \cdot \text{m/s}^2} \right) = \mathbf{1825 \text{ N}}$$



**Discussion** Note the unity conversion factor in the above equation.

---

## 1-39

**Solution** We are to calculate the volume flow rate and mass flow rate of water.

**Assumptions** 1 The volume flow rate, temperature, and density of water are constant over the measured time.

**Properties** The density of water at 15°C is  $\rho = 999.1 \text{ kg/m}^3$ .

**Analysis** The volume flow rate is equal to the volume per unit time, i.e.,

$$\dot{V} = \frac{V}{\Delta t} = \frac{1.5 \text{ L}}{2.85 \text{ s}} \left( \frac{60 \text{ s}}{1 \text{ min}} \right) = 31.58 \text{ L/min} \cong \mathbf{31.6 \text{ Lpm}}$$

where we give our final answer to 3 significant digits, but retain 4 digits to avoid round-off error in the second part of the problem. Since density is mass per unit volume, mass flow rate is equal to volume flow rate times density. Thus,

$$\dot{m} = \rho \dot{V} = (999.1 \text{ kg/m}^3)(31.58 \text{ L/min}) \left( \frac{1 \text{ min}}{60 \text{ s}} \right) \left( \frac{1 \text{ m}^3}{1000 \text{ L}} \right) = \mathbf{0.526 \text{ kg/s}}$$

**Discussion** We used one unity conversion ratio in the first calculation, and two in the second. If we were interested only in the mass flow rate, we could have eliminated the intermediate calculation by solving for mass flow rate directly, i.e.,

$$\dot{m} = \rho \frac{V}{\Delta t} = (999.1 \text{ kg/m}^3) \frac{1.5 \text{ L}}{2.85 \text{ s}} \left( \frac{1 \text{ m}^3}{1000 \text{ L}} \right) = 0.526 \text{ kg/s}$$


---

## 1-40

**Solution** We are to estimate the work and power required to lift a crate.

**Assumptions** 1 The vertical speed of the crate is constant.

**Properties** The gravitational constant is taken as  $g = 9.807 \text{ m/s}^2$ .

**Analysis** (a) Work  $W$  is a form of energy, and is equal to force times distance. Here, the force is the weight of the crate, which is  $F = mg$ , and the vertical distance is  $\Delta z$ , where  $z$  is the elevation.

$$\begin{aligned} W &= F\Delta z = mg\Delta z \\ &= (90.5 \text{ kg})(9.807 \text{ m/s}^2)(1.80 \text{ m}) \left( \frac{1 \text{ N}}{1 \text{ kg} \cdot \text{m/s}^2} \right) \left( \frac{1 \text{ kJ}}{1000 \text{ N} \cdot \text{m}} \right) = 1.5976 \text{ kJ} \cong \mathbf{1.60 \text{ kJ}} \end{aligned}$$

where we give our final answer to 3 significant digits, but retain 5 digits to avoid round-off error in part (b).

(b) Power is work (energy) per unit time. Assuming a constant speed,

$$\dot{W} = \frac{W}{\Delta t} = \frac{1.5976 \text{ kJ}}{12.3 \text{ s}} \left( \frac{1000 \text{ W}}{1 \text{ kJ/s}} \right) = 129.88 \text{ W} \cong \mathbf{130 \text{ W}}$$

Again we give our final answer to 3 significant digits.

**Discussion** The actual required power will be greater than calculated here, due to frictional losses and other inefficiencies in the forklift system. Three unity conversion ratios are used in the above calculations.

---

## 1-41

**Solution** A gas tank is being filled with gasoline at a specified flow rate. Based on unit considerations alone, a relation is to be obtained for the filling time.

**Assumptions** Gasoline is an incompressible substance and the flow rate is constant.

**Analysis** The filling time depends on the volume of the tank and the discharge rate of gasoline. Also, we know that the unit of time is 'seconds'. Therefore, the independent quantities should be arranged such that we end up with the unit of seconds. Putting the given information into perspective, we have

$$t [\text{s}] \leftrightarrow V [\text{L}], \text{ and } \dot{V} [\text{L/s}]$$

It is obvious that the only way to end up with the unit "s" for time is to divide the tank volume by the discharge rate. Therefore, the desired relation is

$$t = \frac{V}{\dot{V}}$$

**Discussion** Note that this approach may not work for cases that involve dimensionless (and thus unitless) quantities.

---

1-42

**Solution** A pool is to be filled with water using a hose. Based on unit considerations, a relation is to be obtained for the volume of the pool.

**Assumptions** Water is an incompressible substance and the average flow velocity is constant.

**Analysis** The pool volume depends on the filling time, the cross-sectional area which depends on hose diameter, and flow velocity. Also, we know that the unit of volume is  $\text{m}^3$ . Therefore, the independent quantities should be arranged such that we end up with the unit of seconds. Putting the given information into perspective, we have

$$\mathcal{V}[\text{m}^3] \text{ is a function of } t[\text{s}], D[\text{m}], \text{ and } V[\text{m/s}]$$

It is obvious that the only way to end up with the unit “ $\text{m}^3$ ” for volume is to multiply the quantities  $t$  and  $V$  with the square of  $D$ . Therefore, the desired relation is

$$\mathcal{V} = CD^2Vt$$

where the constant of proportionality is obtained for a round hose, namely,  $C = \pi/4$  so that  $\mathcal{V} = (\pi D^2/4)Vt$ .

**Discussion** Note that the values of dimensionless constants of proportionality cannot be determined with this approach.

---

## 1-43

**Solution** It is to be shown that the power needed to accelerate a car is proportional to the mass and the square of the velocity of the car, and inversely proportional to the time interval.

**Assumptions** The car is initially at rest.

**Analysis** The power needed for acceleration depends on the mass, velocity change, and time interval. Also, the unit of power  $\dot{W}$  is watt, W, which is equivalent to

$$W = \text{J/s} = \text{N}\cdot\text{m/s} = (\text{kg}\cdot\text{m/s}^2)\text{m/s} = \text{kg}\cdot\text{m}^2/\text{s}^3$$

Therefore, the independent quantities should be arranged such that we end up with the unit  $\text{kg}\cdot\text{m}^2/\text{s}^3$  for power. Putting the given information into perspective, we have

$$\dot{W} [\text{kg}\cdot\text{m}^2/\text{s}^3] \text{ is a function of } m [\text{kg}], V [\text{m/s}], \text{ and } t [\text{s}]$$

It is obvious that the only way to end up with the unit “ $\text{kg}\cdot\text{m}^2/\text{s}^3$ ” for power is to multiply mass with the square of the velocity and divide by time. Therefore, the desired relation is

$$\dot{W} \text{ is proportional to } mV^2/t$$

or,

$$\dot{W} = CmV^2/t$$

where  $C$  is the dimensionless constant of proportionality (whose value is  $1/2$  in this case).

**Discussion** Note that this approach cannot determine the numerical value of the dimensionless numbers involved.

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**Modeling and Solving Engineering Problems**

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**1-44C**

**Solution** We are to discuss the importance of modeling in engineering.

**Analysis** *Modeling* makes it possible to **predict the course of an event before it actually occurs, or to study various aspects of an event mathematically without actually running expensive and time-consuming experiments.** When preparing a mathematical model, all the variables that affect the phenomena are identified, reasonable assumptions and approximations are made, and the interdependence of these variables is studied. The relevant physical laws and principles are invoked, and the problem is formulated mathematically. Finally, the problem is solved using an appropriate approach, and the results are interpreted.

**Discussion** In most cases of actual engineering design, the results are verified by experiment – usually by building a prototype. CFD is also being used more and more in the design process.

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**1-45C**

**Solution** We are to discuss the difference between analytical and experimental approaches.

**Analysis** The *experimental approach (testing and taking measurements)* has the advantage of dealing with the actual physical system, and getting a physical value within the limits of experimental error. However, this approach is expensive, time consuming, and often impractical. The *analytical approach (analysis or calculations)* has the advantage that it is fast and inexpensive, but the results obtained are subject to the accuracy of the assumptions and idealizations made in the analysis.

**Discussion** Most engineering designs require both analytical and experimental components, and both are important. Nowadays, computational fluid dynamics (CFD) is often used in place of pencil-and-paper analysis and/or experiments.

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## 1-46C

**Solution** We are to discuss choosing a model.

**Analysis** The right choice between a crude and complex model is usually **the simplest model that yields adequate results**. Preparing very accurate but complex models is not necessarily a better choice since such models are not much use to an analyst if they are very difficult and time consuming to solve. At a minimum, the model should reflect the essential features of the physical problem it represents. After obtaining preliminary results with the simpler model and optimizing the design, the complex, expensive model may be used for the final prediction.

**Discussion** Cost is always an issue in engineering design, and “adequate” is often determined by cost.

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## 1-47C

**Solution** We are to discuss the difference between accuracy and precision.

**Analysis** *Accuracy* refers to the **closeness of the measured or calculated value to the true value** whereas *precision* represents the **number of significant digits or the closeness of different measurements of the same quantity to each other**. **A measurement or calculation can be very precise without being very accurate, and vice-versa**. When measuring the boiling temperature of pure water at standard atmospheric conditions (100.00°C), for example, a temperature measurement of 97.861°C is very precise, but not as accurate as the less precise measurement of 99.0°C.

**Discussion** Accuracy and precision are often confused; both are important for quality engineering measurements.

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## 1-48C

**Solution** We are to discuss how differential equations arise in the study of a physical problem.

**Analysis** The description of most scientific problems involves equations that relate the *changes* in some key variables to each other, and the smaller the increment chosen in the changing variables, the more accurate the description. In **the limiting case of infinitesimal changes in variables**, we obtain *differential equations*, which provide precise mathematical formulations for the physical principles and laws by representing the rates of changes as *derivatives*.

**Discussion** As we shall see in later chapters, the differential equations of fluid mechanics are known, but very difficult to solve except for very simple geometries. Computers are extremely helpful in this area.

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## 1-49C

**Solution** We are to discuss the value of engineering software packages.

**Analysis** *Software packages are of great value in engineering practice, and engineers today rely on software packages to solve large and complex problems quickly, and to perform optimization studies efficiently.* Despite the convenience and capability that engineering software packages offer, they are still just *tools*, and they cannot replace traditional engineering courses. They simply cause a shift in emphasis in the course material from mathematics to physics.

**Discussion** While software packages save us time by reducing the amount of number-crunching, we must be careful to understand how they work and what they are doing, or else incorrect results can occur.

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## 1-50



**Solution** We are to solve a system of 3 equations with 3 unknowns using appropriate software.

**Analysis** Using EES software, copy the following lines and paste on a blank EES screen to verify the solution:

$$\begin{aligned} 2*x-y+z &= 9 \\ 3*x^2+2*y &= z+2 \\ x*y+2*z &= 14 \end{aligned}$$

**Answers:**  $x = 1.556$ ,  $y = 0.6254$ ,  $z = 6.513$

**Discussion** To obtain the solution in EES, click on the icon that looks like a calculator, or Calculate-Solve.

---

## 1-51



**Solution** We are to solve a system of 2 equations and 2 unknowns using appropriate software.

**Analysis** Using EES software, copy the following lines and paste on a blank EES screen to verify the solution:

$$\begin{aligned} x^3-y^2 &= 10.5 \\ 3*x*y+y &= 4.6 \end{aligned}$$

**Answers:**  $x = 2.215$ ,  $y = 0.6018$

**Discussion** To obtain the solution in EES, click on the icon that looks like a calculator, or Calculate-Solve.

---

1-52 

**Solution** We are to determine a positive real root of the following equation using appropriate software:

$$3.5x^3 - 10x^{0.5} - 3x = -4.$$

**Analysis** Using EES software, copy the following lines and paste on a blank EES screen to verify the solution:

$$3.5*x^3-10*x^{0.5}-3*x = -4$$

**Answer:**  $x = 1.554$

**Discussion** To obtain the solution in EES, click on the icon that looks like a calculator, or [Calculate-Solve](#).

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1-53 

**Solution** We are to solve a system of 3 equations with 3 unknowns using appropriate software.

**Analysis** Using EES software, copy the following lines and paste on a blank EES screen to verify the solution:

$$x^2*y-z=1.5$$

$$x-3*y^{0.5}+x*z=-2$$

$$x+y-z=4.2$$

**Answers:**  $x = 0.9149$ ,  $y = 10.95$ ,  $z = 7.665$

**Discussion** To obtain the solution in EES, click on the icon that looks like a calculator, or [Calculate-Solve](#).

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**Review Problems**

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**1-54E**

**Solution** We are to estimate the rate of heat transfer into a room and the cost of running an air conditioner for one hour.

**Assumptions** 1 The rate of heat transfer is constant. 2 The indoor and outdoor temperatures do not change significantly during the hour of operation.

**Analysis**

(a) In one hour, the air conditioner supplies 5,000 Btu of cooling, but runs only 60% of the time. Since the indoor and outdoor temperatures remain constant during the hour of operation, the average rate of heat transfer into the room is the same as the average rate of cooling supplied by the air conditioner. Thus,

$$\dot{Q} = \frac{0.60(5000 \text{ Btu})}{1 \text{ h}} = \mathbf{3,000 \text{ Btu/h}} \left( \frac{1 \text{ kW}}{3412.14 \text{ Btu/h}} \right) = \mathbf{0.879 \text{ kW}}$$

(b) Energy efficiency ratio is defined as the amount of heat removed from the cooled space in Btu for 1 Wh (watt-hour) of electricity consumed. Thus, for every Wh of electricity, this particular air conditioner removes 9.0 Btu from the room. To remove 3,000 Btu in one hour, the air conditioner therefore consumes  $3,000/9.0 = 333.33 \text{ Wh} = 0.33333 \text{ kWh}$  of electricity. At a cost of 7.5 cents per kWh, it costs only **2.50 cents** to run the air conditioner for one hour.

**Discussion** Notice the unity conversion ratio in the above calculation. We also needed to use some common sense and dimensional reasoning to come up with the appropriate calculations. While this may seem very cheap, if this air conditioner is run at these conditions continuously for one month, the electricity will cost  $(\$0.025/\text{h})(24 \text{ h/day})(30 \text{ day/mo}) = \$18/\text{mo}$ .

---

## 1-55

**Solution** The gravitational acceleration changes with altitude. Accounting for this variation, the weights of a body at different locations are to be determined.

**Analysis** The weight of an 65-kg man at various locations is obtained by substituting the altitude  $z$  (values in m) into the relation

$$W = mg = (65 \text{ kg})(9.807 - 3.32 \times 10^{-6} z) \left( \frac{1 \text{ N}}{1 \text{ kg} \cdot \text{m/s}^2} \right) \quad (\text{where } z \text{ is in units of m/s}^2)$$

Sea level:  $(z = 0 \text{ m}): W = 65 \times (9.807 - 3.32 \times 10^{-6} \times 0) = 65 \times 9.807 = \mathbf{637.5 \text{ N}}$

Denver:  $(z = 1610 \text{ m}): W = 65 \times (9.807 - 3.32 \times 10^{-6} \times 1610) = 65 \times 9.802 = \mathbf{637.1 \text{ N}}$

Mt. Ev.:  $(z = 8848 \text{ m}): W = 65 \times (9.807 - 3.32 \times 10^{-6} \times 8848) = 65 \times 9.778 = \mathbf{635.5 \text{ N}}$

**Discussion** We report 4 significant digits since the values are so close to each other. The percentage difference in weight from sea level to Mt. Everest is only about  $-0.3\%$ , which is negligible for most engineering calculations.

---

## 1-56E

**Solution** The thrust developed by the jet engine of a Boeing 777 is given to be 85,000 pounds. This thrust is to be expressed in N and kgf.

**Analysis** Noting that  $1 \text{ lbf} = 4.448 \text{ N}$  and  $1 \text{ kgf} = 9.81 \text{ N}$ , the thrust developed is expressed in two other units as

$$\text{Thrust in N:} \quad \text{Thrust} = (85,000 \text{ lbf}) \left( \frac{4.448 \text{ N}}{1 \text{ lbf}} \right) = \mathbf{3.78 \times 10^5 \text{ N}}$$

$$\text{Thrust in kgf:} \quad \text{Thrust} = (37.8 \times 10^5 \text{ N}) \left( \frac{1 \text{ kgf}}{9.81 \text{ N}} \right) = \mathbf{3.85 \times 10^4 \text{ kgf}}$$



**Discussion** Because the gravitational acceleration on earth is close to  $10 \text{ m/s}^2$ , it turns out that the two force units N and kgf differ by nearly a factor of 10. This can lead to confusion, and we recommend that you do not use the unit kgf.

---

## 1-57

**Solution** The constants appearing dynamic viscosity relation for methanol are to be determined using the data in Table A-7.

**Analysis** Using the data from Table A-7, we have

$$\begin{aligned} 5.857 \times 10^{-4} &= a10^{b/(293-c)} \\ 4.460 \times 10^{-4} &= a10^{b/(313-c)} \\ 3.510 \times 10^{-4} &= a10^{b/(333-c)} \end{aligned}$$

which is a nonlinear system of three algebraic equations. Using EES or any other computer code, one finds

$$a = 8.493 \times 10^{-6} \text{ Pa}\cdot\text{s} \quad b = 534.5 \text{ k} \quad c = 2.27 \text{ K}$$

Then the viscosity correlation for methanol becomes

$$\mu = (8.493 \times 10^{-6}) \times 10^{534.5/(T-2.27)}$$

For  $T = 50^\circ\text{C} = 323 \text{ K}$  the correlation gives  $\mu = 3.941 \times 10^{-4} \text{ Pa}\cdot\text{s}$ , which is nicely agreeing with the data in Table A-7.

---

## 1-58

**Solution** A relation for the terminal settling velocity of a solid particle is given. The dimension of a parameter in the relation is to be determined.

**Analysis** We have the dimensions for each term except  $F_L$ .

$$\begin{aligned} [g] &= [LT^{-2}] \\ [D] &= [L] \\ [V_L] &= [LT^{-1}] \end{aligned}$$

and

$$[LT^{-1}] = [F_L][2LT^{-2}]^{1/2}[L]^{1/2} = \sqrt{2}[F_L][LT^{-1}]$$

Therefore  $F_L$  is a dimensionless coefficient that is this equation is dimensionally homogeneous, and should hold for any unit system.

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## 1-59

**Solution** The flow of air through a wind turbine is considered. Based on unit considerations, a proportionality relation is to be obtained for the mass flow rate of air through the blades.

**Assumptions** Wind approaches the turbine blades with a uniform velocity.

**Analysis** The mass flow rate depends on the air density, average wind velocity, and the cross-sectional area which depends on hose diameter. Also, the unit of mass flow rate  $\dot{m}$  is kg/s. Therefore, the independent quantities should be arranged such that we end up with the proper unit. Putting the given information into perspective, we have

$$\dot{m} \text{ [kg/s]} \text{ is a function of } \rho \text{ [kg/m}^3\text{]}, D \text{ [m]}, \text{ and } V \text{ [m/s]}$$

It is obvious that the only way to end up with the unit “kg/s” for mass flow rate is to multiply the quantities  $\rho$  and  $V$  with the square of  $D$ . Therefore, the desired proportionality relation is

$$\dot{m} \text{ is proportional to } \rho D^2 V$$

or,

$$\dot{m} = C \rho D^2 V$$

where the constant of proportionality is  $C = \pi/4$  so that  $\dot{m} = \rho(\pi D^2/4)V$

**Discussion** Note that the dimensionless constants of proportionality cannot be determined with this approach.

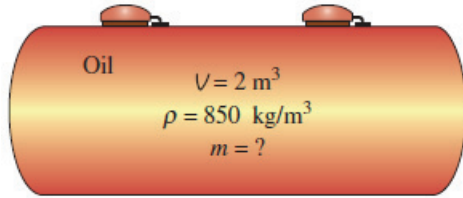
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1-60

**Solution** The volume of an oil tank is given. The mass of oil is to be determined.

**Assumptions** Oil is a nearly incompressible substance and thus its density is constant.

**Analysis** A sketch of the system is given below.



Suppose we forgot the formula that relates mass to density and volume. However, we know that mass has the unit of kilograms. That is, whatever calculations we do, we should end up with the unit of kilograms. Putting the given information into perspective, we have

$$\rho = 850 \text{ kg/m}^3 \text{ and } V = 2 \text{ m}^3$$

It is obvious that we can eliminate  $\text{m}^3$  and end up with kg by multiplying these two quantities. Therefore, the formula we are looking for should be

$$m = \rho V$$

Thus,

$$m = (850 \text{ kg/m}^3)(2 \text{ m}^3) = \mathbf{1700 \text{ kg}}$$

**Discussion** Note that this approach may not work for more complicated formulas. Nondimensional constants also may be present in the formulas, and these cannot be derived from unit considerations alone.

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Fundamentals of Engineering (FE) Exam Problems

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**1-61**

If mass, heat, and work are not allowed to cross the boundaries of a system, the system is called

- (a) Isolated      (b) Isothermal      (c) Adiabatic      (d) Control mass      (e) Control volume

Answer (a) Isolated

**1-62**

The speed of an aircraft is given to be 260 m/s in air. If the speed of sound at that location is 330 m/s, the flight of aircraft is

- (a) Sonic      (b) Subsonic      (c) Supersonic      (d) Hypersonic

Answer (b) Subsonic

**1-63**

One J/kg is equal to

- (a)  $1 \text{ kPa}\cdot\text{m}^3$       (b)  $1 \text{ kN}\cdot\text{m}/\text{kg}$       (c)  $0.001 \text{ kJ}$       (d)  $1 \text{ N}\cdot\text{m}$       (e)  $1 \text{ m}^2/\text{s}^2$

Answer: (e)  $1 \text{ m}^2/\text{s}^2$

**1-64**

Which is a unit for power?

- (a) Btu      (b) kWh      (c) kcal      (d) hph      (e) kW

Answer: (e) kW

## 1-65

The speed of an aircraft is given to be 950 km/h. If the speed of sound at that location is 315 m/s, the Mach number is

- (a) 0.63      (b) 0.84      (c) 1.0      (d) 1.07      (e) 1.20

*Answer* (b) 0.84

**Solution** Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

```
Vel=950 [km/h]*Convert(km/h, m/s)
c=315 [m/s]
Ma=Vel/c
```

## 1-66

The weight of a 10-kg mass at sea level is

- (a) 9.81 N      (b) 32.2 kgf      (c) 98.1 N      (d) 10 N      (e) 100 N

*Answer* (c) 98.1 N

**Solution** Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

```
m=10 [kg]
g=9.81 [m/s^2]
W=m*g
```

## 1-67

The weight of a 1-lbm mass is

- (a)  $1 \text{ lbm}\cdot\text{ft}/\text{s}^2$     (b) 9.81 lbf    (c) 9.81 N    (d) 32.2 lbf    (e) 1 lbf

*Answer* (e) 1 lbf

**Solution** Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

```
m=1 [lbm]
g=32.2 [ft/s^2]
W=m*g*Convert(lbm-ft/s^2, lbf)
```

## 1-68

A hydroelectric power plant operates at its rated power of 12 MW. If the plant has produced 26 million kWh of electricity in a specified year, the number of hours the plant has operated that year is

- (a) 2167 h    (b) 2508 h    (c) 3086 h    (d) 3710 h    (e) 8760 h

*Answer* (a) 2167 h

**Solution** Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

```
RatedPower=12000 [kW]
ElectricityProduced=26E6 [kWh]
Hours=ElectricityProduced/RatedPower
```

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**Design and Essay Problems**


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## 1-69 to 1-72

**Solution** Students' essays and designs should be unique and will differ from each other.

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